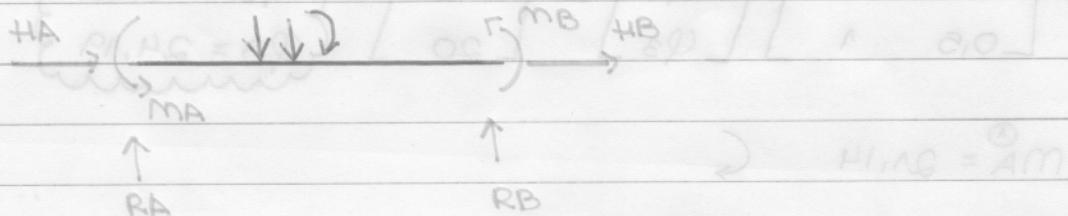


# Método dos deslocamentos



$$M_A = M_{AB} + \beta_{AB}(Q_A + \alpha_{BA} \cdot \beta_{BA} \cdot Q_B) + \beta_{AB} (1 + \alpha_{AB}) (V_A - V_B)$$

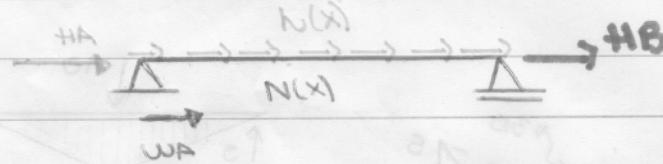
$$M_B = M_{BA} + \beta_{BA} \cdot \alpha_{AB} \cdot (Q_A + \beta_{BA} \cdot Q_B) + \beta_{BA} (1 + \alpha_{BA}) (V_A - V_B)$$

$$R_A = R_{AB} + (M_A - M_{AB} + M_B - M_{BA})$$

$$R_B = R_{BA} - (M_A - M_{AB} + M_B - M_{BA})$$

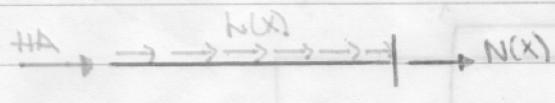
$$H_A = \varphi_A(H_{AB}, W_A, W_B)$$

$$H_B = \varphi_B (H_{BA}, W_A, W_B)$$



$$H_A = - \int_0^L n(x) dx - H_B$$

$$N(x) = -H_A - \int_0^x n(t) dt$$

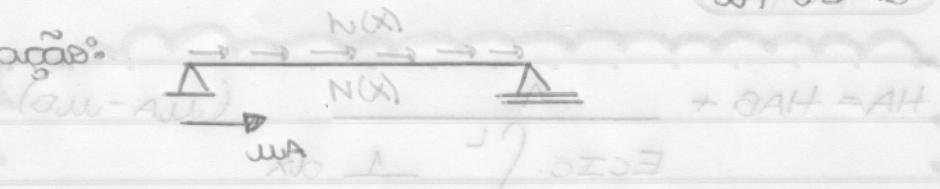


## Páginas de uso

PTV: Estado de Carregamento:

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Estados de deformações:



$+ \partial AH = AH$

"modo"

$\Delta V_{ext} = \bar{w}_B - \bar{w}_A$

$\Delta V_{int} = \int_0^L 1 \times N(x) \, dx$

$\bar{w}_B - \bar{w}_A = \int_0^L \frac{N(x)}{EA} \, dx$

$N(x) = -HA - \int_0^x h(t) \, dt$

$(\bar{w}_B - \bar{w}_A) = HB + \int_0^L h(x) \, dx - AH = AH$

$\int_0^x h(t) \, dt$

 $\rightarrow HT \rightarrow$  integral detodas as forças internas atuando sobre  
a viga  $+ AH = BH$ 

$w_B = w_A + Ec Ic \int_0^L \frac{N(x)}{EA} \, dx$

$w_B = w_A + Ec Ic \int_0^L \frac{(HB + HT - \int_0^x h(t) \, dt)}{EA} \, dx$

$w_B = w_A + Ec Ic \cdot (HB + HT) \int_0^L \frac{1}{EA} \, dx - Ec Ic \int_0^L \frac{\int_0^x h(t) \, dt}{EA} \, dx$

$\bullet HB = HBA + \frac{1}{EA} (\bar{w}_B - \bar{w}_A)$  "Global"

para vigas prismáticas:

$HB = HBA + \frac{1}{EA} (\bar{w}_B - \bar{w}_A)$

$* HA + HB + HT = 0$

$HA = -HB - HT$

$$HA = \left( \frac{-HBA - HT}{L \cdot Ic Ic} \right) - \frac{1}{Ec Ic} \int_0^L \frac{1}{EA} \, dx = \frac{(\bar{w}_B - \bar{w}_A) \cdot Ad}{Ec Ic} = \text{seu valor}$$

↓  
Chamam de  $HAB$

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$$HA = HAB + \frac{1}{ECIC} \int_0^L \frac{1}{EA} dx \quad (\mu_A - \mu_B)$$

$$ECIC \int_0^L \frac{1}{EA} dx$$

"GLOBAL"

$$A\bar{U} - \partial\bar{U} = \partial VT$$

$$x_b \bar{U}_B \times 1 = \partial VT$$

$$HA = HAB + EA \int_0^L \frac{1}{EA} dx \quad (\mu_A - \mu_B) \quad \text{para vigas pneumáticas}$$

$$L \cdot ECIC$$

Resumindo:

$$HA = HAB + \frac{1}{EA} \int_0^L \frac{1}{EA} dx \quad (\mu_A - \mu_B)$$

$$ECIC \int_0^L \frac{1}{EA} dx$$

$$HB = HBA + \frac{1}{EA} \int_0^L \frac{1}{EA} dx \quad (\mu_B - \mu_A)$$

$$ECIC \int_0^L \frac{1}{EA} dx$$

$$x_b \bar{U}_B \times 1 = A\bar{U} - \partial\bar{U}$$

$$A\bar{U} - \partial\bar{U}$$

$$x_b \bar{U}_B \times 1 = A\bar{U} - \partial\bar{U}$$

$$A\bar{U} - \partial\bar{U}$$

$$x_b \bar{U}_B \times 1 = A\bar{U} - \partial\bar{U}$$

$$A\bar{U} - \partial\bar{U}$$

HA	HAB	$\frac{1}{EA} \int_0^L \frac{1}{EA} dx$	$\frac{1}{EA} \int_0^L \frac{1}{EA} dx$	$\frac{1}{EA} \int_0^L \frac{1}{EA} dx$	UA
RA	RAB	0	$K_{22} \frac{\beta_{AB}(1+\alpha_{AB})}{L}$	$-K_{22} \frac{\beta_{BA}(1+\alpha_{BA})}{L}$	VA
MA	MAB	0	$\beta_{AB}(1+\alpha_{AB})$	$\beta_{BA}(1+\alpha_{BA})$	UA
HB	HBA	$\frac{1}{EA} \int_0^L \frac{1}{EA} dx$	0	$\frac{1}{EA} \int_0^L \frac{1}{EA} dx$	UB
RB	RBA	0	$-K_{22} \frac{\beta_{AB}(1+\alpha_{AB})}{L}$	$K_{22} \frac{\beta_{BA}(1+\alpha_{BA})}{L}$	VB
MB	MBA	0	$\beta_{BA}(1+\alpha_{BA})$	$\beta_{BA} \cdot \alpha_{BA}$	UB

$$0 = T_4 + \partial U - A_4$$

$$* K_{22} = \frac{\beta_{AB}(1+\alpha_{AB}) + \beta_{BA}(1+\alpha_{BA})}{L^2}$$

$$\frac{1}{EA} \int_0^L \frac{1}{EA} dx = \frac{1}{EA} \int_0^L \frac{1}{EA} dx$$

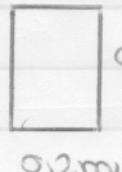
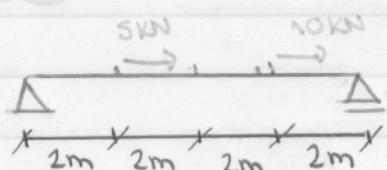
componendo

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para vigas promáticas: matriz simétrica [K]

HA	HAB	<u>EA</u> LECIC	0	0	-EA LECIC	0	0	WA
RA	RAB	0	<u>12</u> <u>L<sup>2</sup>L'</u>	<u>6</u> <u>LL'</u>	0	<u>-12</u> <u>L<sup>2</sup>L'</u>	<u>6</u> <u>LL'</u>	NA
MA	MAB	0	<u>6</u> <u>LL'</u>	<u>4</u> <u>L'</u>	0	<u>-6</u> <u>LL'</u>	<u>2</u> <u>L'</u>	UA
=								
HB	HBA	<u>-EA</u> LECIC	0	0	<u>EA</u> LECIC	0	0	WB
RB	RBA	0	<u>-12</u> <u>L<sup>2</sup>L'</u>	<u>-6</u> <u>LL'</u>	0	<u>12</u> <u>L<sup>2</sup>L'</u>	<u>-6</u> <u>LL'</u>	NB
MB	MBA	0	<u>6</u> <u>LL'</u>	<u>2</u> <u>L'</u>	0	<u>-6</u> <u>LL'</u>	<u>4</u> <u>L'</u>	UB

Exercício: Cálculo das rigidez axial e de HAB, HBA



0.6m

0.2m

$$E = 25000 \text{ MPa} = 25 \times 10^6 \text{ N/mm}^2$$

$$Ecic = 10^5 \text{ kN.m}^2$$

① Caracterizações da viga

$$\text{Rigidez Axial} = \frac{1}{Ecic} = \frac{1}{\frac{1}{EA}} = EA = \frac{30}{8} = 3.75 \text{ kN/mm}$$

$$EA = 25 \cdot 10^6 \cdot 0.12 \cdot 0.15 = 3 \cdot 10^6 \text{ N/mm}$$

$$EI = 25 \cdot 10^6 \cdot \frac{0.12 \cdot 0.15^3}{12} = 90 \cdot 10^3 \text{ N/mm}$$

$$L' = 8$$

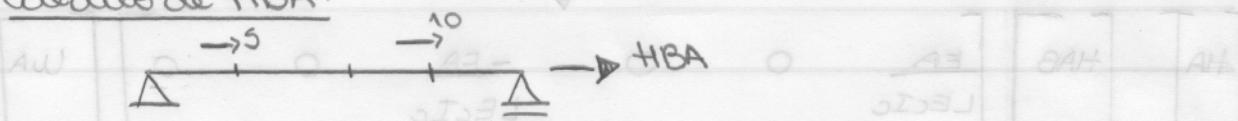
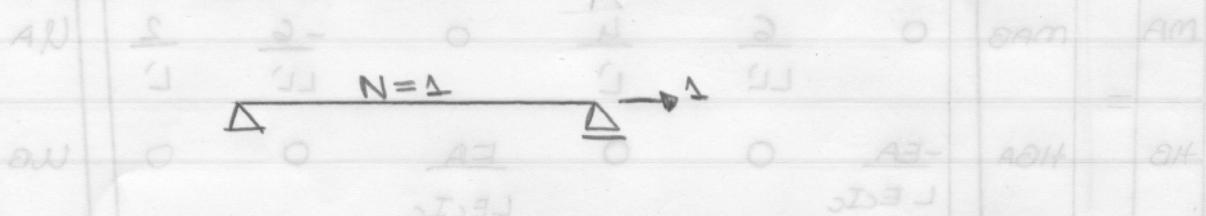
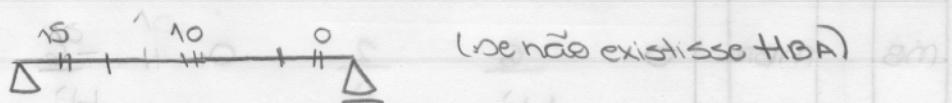
$$* L' Ecic = \frac{8 \cdot 10^5}{9 \cdot 10^4} = 8.8$$

$$* \beta = \frac{4}{8.8} = 0.45$$

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Exercício de sistema estrutural com apoio móvel

\* Cálculo de HBA:

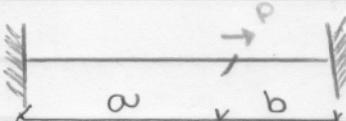
↳ Estados de carregamentos:↳ Estados de Deformações:

$$\delta_{\Delta 0} = \int_0^8 \frac{du}{dx} \cdot N dx = \int_0^2 \frac{15}{EA} dx + \int_2^6 \frac{10}{EA} dx = \frac{70}{EA}$$

$$\delta_{\Delta \Delta} = \int_0^8 \frac{1}{EA} dx = \frac{8}{EA}$$

$$\delta_1 = \delta_{\Delta 0} + HBA \cdot \delta_m = 0 \Rightarrow HBA = -\frac{70}{8} = -8,75 \quad (+HBA = -8,75)$$

$$HAB + HBA + 15 = 0 \quad \{ HAB = -6,25 \}$$

• Intuitivamente

$$HAB = -P \cdot b \quad HBA = -P \cdot a$$

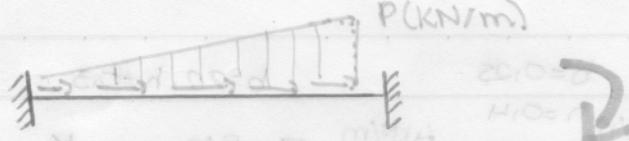
Conferindo os exercícios

$$HAB = -\frac{5 \cdot 6}{8} - 10 \cdot 2 = -\frac{50}{8} = -6,25 \quad \} \text{OK!}$$

$$HBA = -\frac{5 \cdot 2}{8} - \frac{10 \cdot 6}{8} = -\frac{70}{8} = -8,75$$

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a) NO NO



$$H_{AB} = -\frac{PL}{12}$$

$$H_{BA} = -\frac{PL}{6}$$

$$6.0 \times d$$

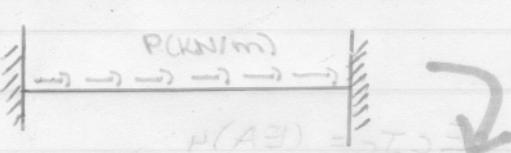
$$6.0 \times 1$$

$$H_{BA}$$

$$5.0 \times d$$

$$5.0 \times 1$$

$$H_{BA}$$



$$H_{AB} = -\frac{PL}{20}$$

$$H_{EA} = -\frac{PL}{2}$$

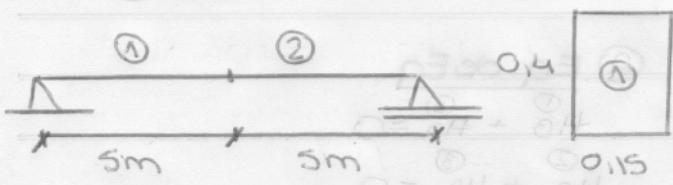
$$(au - av) \quad AE + AH = AH$$

$$\Delta EJL$$

$$(av - au) \quad AE + AH = AH$$

$$\Delta EJL$$

**Exercício** Calcular rigidez axial.



$$A_1 = 0,06$$

$$EA_1 = 1,5 \cdot 10^6 \text{ KN}$$

$$E = 25 \cdot 10^6 \text{ KN/m}^2$$

$$[2] \quad 0,6 \quad ECIC = 10^5 \text{ KNm}^2$$

$$A_2 = 0,12$$

$$EA_2 = 3 \cdot 10^6 \text{ KN}$$

$$\text{*Rigidez Axial: } \frac{1}{ECIC \int \frac{1}{EA} dx} = \frac{1}{ECIC \left( \frac{5}{EA_1} + \frac{5}{EA_2} \right)}$$

$$= \frac{1}{5 \left( \frac{1}{15} + \frac{1}{30} \right)} = \frac{1}{\frac{5}{30}} = 2$$

$$\text{or. med. 2 = AE } m^2 = 1 \text{ [eqd]}$$

$$\text{or. med. 2 = AE } m^2 = 1 \text{ [eqd]}$$

$$\text{or. med. 2 = AE } m^2 = 1 \text{ [eqd]}$$

$$m^2 \cdot 0,1 = AE \quad m^2 = 1 \text{ [eqd]}$$

$$m^2 = AH$$

$$m^2 = AH$$